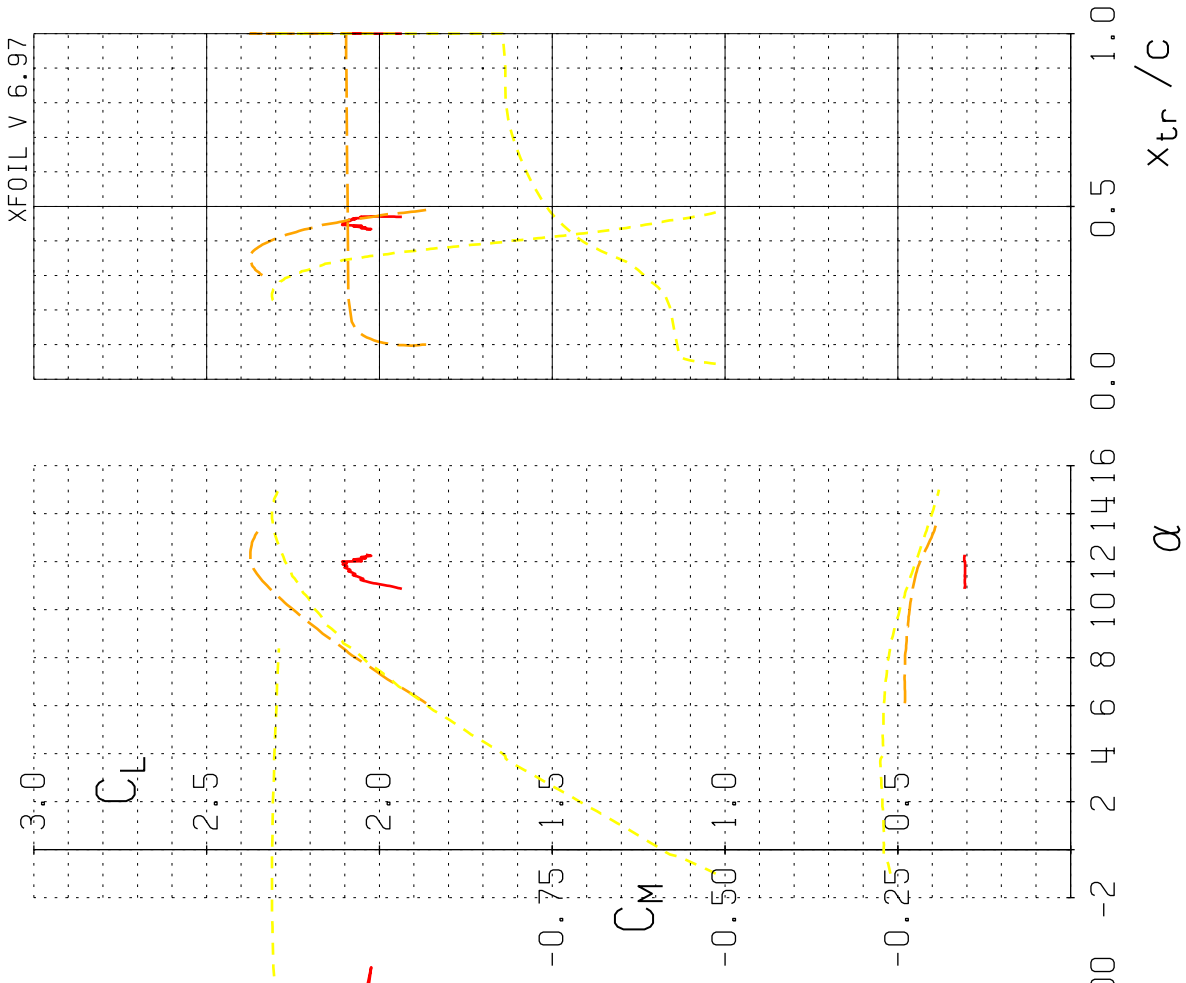
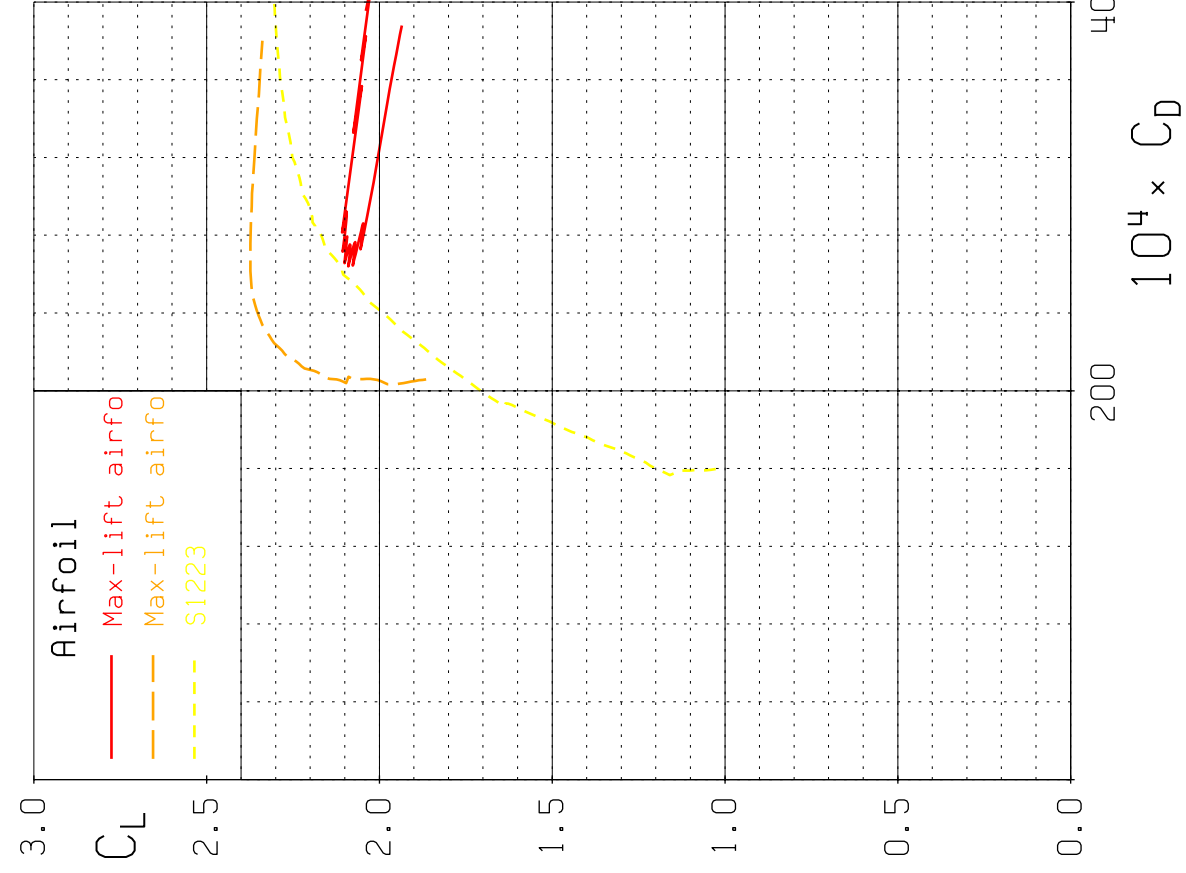


Max-lift airfoil 5 Re = 300000 Ma = 0.000 Ncrit = 9.000
 Max-lift airfoil 5s Re = 300000 Ma = 0.000 Ncrit = 9.000
 S1223 Re = 300000 Ma = 0.000 Ncrit = 9.000



x_{tr} / c